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APPLICATION NO.	FILING DATE	FIRST NAMED INVENTOR	ATTORNEY DOCKET NO.	CONFIRMATION NO.
09/700,316	11/14/2000	Lars-Olof Ohberg	1878/00037	4171
EDWARD A. P	7590 10/17/2007 PENNINGTON		EXAM	INER
SWIDLER BERLIN SHEREFF FRIEDMAN, LLP			LEE, BENJAMIN WILLIAM	
3000 K STREE' SUITE 300	Τ		ART UNIT	PAPER NUMBER
WASHINGTON	N, DC 20007		3714	
			MAIL DATE	DELIVERY MODE
		•	10/17/2007	PAPER

Please find below and/or attached an Office communication concerning this application or proceeding.

The time period for reply, if any, is set in the attached communication.

•	HT.	
	Application No.	Applicant(s)
	09/700,316	OHBERG ET AL.
Office Action Summary	Examiner	Art Unit
	Benjamin W. Lee	3714
The MAILING DATE of this communication app Period for Reply	ears on the cover sheet with the c	orrespondence address
A SHORTENED STATUTORY PERIOD FOR REPLY WHICHEVER IS LONGER, FROM THE MAILING DA - Extensions of time may be available under the provisions of 37 CFR 1.13 after SIX (6) MONTHS from the mailing date of this communication. - If NO period for reply is specified above, the maximum statutory period w - Failure to reply within the set or extended period for reply will, by statute, Any reply received by the Office later than three months after the mailing earned patent term adjustment. See 37 CFR 1.704(b).	ATE OF THIS COMMUNICATION 36(a). In no event, however, may a reply be tin vill apply and will expire SIX (6) MONTHS from , cause the application to become AB ANDONE	N. nely filed the mailing date of this communication. D (35 U.S.C. § 133).
Status		1
 1) Responsive to communication(s) filed on 26 Jule 2a) This action is FINAL. 2b) This 3) Since this application is in condition for allower closed in accordance with the practice under E 	action is non-final. nce except for formal matters, pro	
•	x parte Quayle, 1935 C.D. 11, 45	03 O.G. 213.
Disposition of Claims		
4) ☐ Claim(s) 18 and 20-30 is/are pending in the ap 4a) Of the above claim(s) is/are withdraw 5) ☐ Claim(s) is/are allowed. 6) ☐ Claim(s) 18 and 20-30 is/are rejected. 7) ☐ Claim(s) is/are objected to. 8) ☐ Claim(s) are subject to restriction and/or	vn from consideration.	
Application Papers	•	
9) The specification is objected to by the Examine 10) The drawing(s) filed on is/are: a) accomplicant may not request that any objection to the Replacement drawing sheet(s) including the correct 11) The oath or declaration is objected to by the Example 11.	epted or b) objected to by the l drawing(s) be held in abeyance. Sed ion is required if the drawing(s) is ob	e 37 CFR 1.85(a). jected to. See 37 CFR 1.121(d).
Priority under 35 U.S.C. § 119		
 12) ☒ Acknowledgment is made of a claim for foreign a) ☒ All b) ☐ Some * c) ☐ None of: 1. ☐ Certified copies of the priority documents 2. ☐ Certified copies of the priority documents 3. ☒ Copies of the certified copies of the priority application from the International Bureau * See the attached detailed Office action for a list 	s have been received. s have been received in Applicati rity documents have been receive u (PCT Rule 17.2(a)).	on No ed in this National Stage
Attachment(s) 1) Notice of References Cited (PTO-892)	4) ☐ Interview Summary	(PTO-413)
2) Notice of Draftsperson's Patent Drawing Review (PTO-948) 3) Information Disclosure Statement(s) (PTO/SB/08) Paper No(s)/Mail Date	Paper No(s)/Mail Di 5) Notice of Informal F 6) Other:	ate

DETAILED ACTION

Continued Examination Under 37 CFR 1.114

1. A request for continued examination under 37 CFR 1.114, including the fee set forth in 37 CFR 1.17(e), was filed in this application after final rejection. Since this application is eligible for continued examination under 37 CFR 1.114, and the fee set forth in 37 CFR 1.17(e) has been timely paid, the finality of the previous Office action has been withdrawn pursuant to 37 CFR 1.114. Applicant's submission filed on 07/26/2007 has been entered. Claims 18 and 20-30 are pending in this application. Claims 18, 23, and 27 have been amended.

Claim Rejections - 35 USC § 103

- 2. The following is a quotation of 35 U.S.C. 103(a) which forms the basis for all obviousness rejections set forth in this Office action:
 - (a) A patent may not be obtained though the invention is not identically disclosed or described as set forth in section 102 of this title, if the differences between the subject matter sought to be patented and the prior art are such that the subject matter as a whole would have been obvious at the time the invention was made to a person having ordinary skill in the art to which said subject matter pertains. Patentability shall not be negatived by the manner in which the invention was made.
- 3. This application currently names joint inventors. In considering patentability of the claims under 35 U.S.C. 103(a), the examiner presumes that the subject matter of the various claims was commonly owned at the time any inventions covered therein were made absent any evidence to the contrary. Applicant is advised of the obligation under 37 CFR 1.56 to point out the inventor and invention dates of each claim that was not commonly owned at the time a later

invention was made in order for the examiner to consider the applicability of 35 U.S.C. 103(c) and potential 35 U.S.C. 102(e), (f) or (g) prior art under 35 U.S.C. 103(a).

4. Claims 18, 20, 21, 23-25, and 27-30 are rejected under 35 U.S.C. 103(a) as being unpatentable over Eldridge (US 5,228,854) in view of Schroeder (US 5,631,830).

Re claims 18, 20, 21, 23-25, and 27-30: Eldridge discloses a method of simulating a missile by means of a missile simulator during testing of an aircraft which includes a weapon system for controlling missiles with which the aircraft may be equipped (see Fig. 4; col. 3, lines 23-39), the method comprising generating a target seeker command position for a simulated target seeker, whereby the simulated target seeker is commanded to adopt a predetermined position (see col. 5, lines 40-44), wherein the simulated target seeker is assumed to move at finite speeds (see col. 6, lines 10-14; col. 9, lines 36-41) and that its movement is constrained to a single plane (see col. 5, lines 55-67), and simulating behavior of the missile in a computer model to generate an actual value signal adapted to the weapon system (see col. 5, lines 55-62; col. 6, lines 55-60). The simulated target seeker is assumed to move at finite speeds because the missile model is updated according to the changing position of the target and the movement of the simulated missile is constrained to a single plane that is perpendicular to the target. Eldridge further discloses computer circuitry operable to run a computer model of a missile (see ref. nos. 172 and 174 in Fig. 4; col. 9, lines 31-36) and interface circuitry communicatively connectable between the computer circuitry and a weapons system of an aircraft (see ref. no. 504 in Fig. 4;

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col. 6, lines 51-68). Eldridge further teaches that the missile may be partially guided by the radar

system of the attack aircraft (see col. 7, lines 3-10).

However, regarding claim 18, Eldridge fails to disclose generating in the weapon system a trouble signal from a deviation between the target seeker command position and the actual value signal, wherein the trouble signal is measured continuously and wherein sampled values for a vector indicating error in amplitude and phase, which represent a difference vector corresponding to the target seeker command position and a vector corresponding to the actual value signal, are determined and sent to the computer model in the missile simulator, using the trouble signal as a control signal for the simulated target seeker, and repeating the control system steps. Regarding claims 23 and 27, Eldridge fails to disclose generating a signal representing a deviation of a simulated target seeker form a commanded position of the simulated target seeker, using the deviation signal in the missile computer model, and sending the actual value signal to the computer model.

Schroeder teaches a conventional missile control system. The conventional missile control system measures the actual value signal/measured dynamic response 205 of the missile and determines a trouble signal/error signal 220 by comparing the dynamic response 205 to a target seeker command position/commanded dynamic response signal 215. The target seeker/autopilot controller 225 then uses the error signal 220 to actuate the control devices of the missile in order to guide the missile towards the target (see Fig. 2; col. 2, lines 6-26).

Therefore, in view of Schroeder, it would have been obvious to one of ordinary skill in the art at the time the invention was made to use the missile control system of Schroeder in the

weapons system of Eldridge in order to provide a realistic simulation of the control system of a conventional missile.

Regarding the limitations added on 07/26/2007 (i.e. "single aircraft" and "in the missile simulator" in claim 18, "using a missile simulator in a single aircraft" in claim 23, and "used in a single aircraft" in claim 27), replacing the target pod of Eldridge with a simulated target and moving the simulation to the attacking plane would have been a simple substitution of one known element for another to obtain predictable results. In the field of simulation and testing, it is well known in the art to test actual components by feeding the actual component simulated inputs. For example, the previously cited Watson reference ("Distributed Simulation Testing for Weapons System Performance of the F/A-18 and AIM-120 AMRAAM") mentions Hardware in the Loop (HWIL) testing, which is a technique used to test embedded systems by feeding simulated inputs to an actual control system and using the outputs in a simulation to generate new inputs. It would have been obvious to one of ordinary skill in the art to use a simulated target instead of an actual target in order to reduce the overall cost of the system (it will require one less plane and simplify the overall operation of the system). Furthermore, simulators in which the simulation occurs on one apparatus instead of two are old and well known in the art.

5. Claims 22 and 26 are rejected under 35 U.S.C. 103(a) as being unpatentable over Eldridge as modified by Schroeder as applied to claims 18 and 23 above, and further in view of Phillips (Feedback Control Systems, 3rd ed.).

The teachings of Eldridge as modified by Schroeder as applied to claims 18 and 23 above have been discussed.

However, the teachings of Eldridge as modified by Schroeder as applied to claims 18 and 23 fail to disclose or fairly suggest a time-continuous actual value signal is reproduced from a time-discrete vector from the computer model.

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Phillips teaches a method of modeling a feedback control system comprising time discrete signals (see p. 468).

Therefore, it would have been obvious to one of ordinary skill in the art at the time the invention was made to have modified the missile control system described by Schroeder by applying a linear time-invariant discrete feedback system, in light of the teachings of Phillips, in order to allow modeling of digital controllers that can accept information only at discrete values of time (see p. 469).

Response to Arguments

Applicant's arguments filed 07/26/2007 have been fully considered but they are not 6. persuasive.

The applicant's arguments regarding the newly added limitations to the claims have been addressed in the body of the rejections above.

The examiner respectfully disagrees with the applicant's argument that neither Eldridge, nor Schroeder, nor the combination of Eldridge and Schroeder teaches receiving the target seeker command position at the weapon system. The examiner is unclear regarding the distinction the applicant makes between "lock-on information" in Eldridge and "target seeker command position." The examiner interprets "lock-on information" to include position information (of the missile and target). It is unclear to the examiner how the determination of whether a missile is

locked-on or not can be made without knowing the respective positions of the missile and target.

Conventional missile guidance systems use control systems which use sensors to determine the actual position of the missile and use a feedback system to correct positioning errors with respect to the intended target location.

The examiner respectfully disagrees with the applicant's argument that neither Eldridge nor Schroeder teach continuously generating a trouble signal (i.e. error signal) based on a continuously measured position vector and a target vector and using the error signal to control a simulated missile. As taught by Schroeder, conventional missiles use such a control system. It would have been obvious to one of ordinary skill in the art to incorporate the technique used by an actual missile into the simulation of a missile. The goal of simulators is to model real-world objects and events as accurately as possible. Using the same control system principles of an actual missile system in a simulator would have been obvious and the results of such a replacement would have been predictable.

Conclusion

7. Any inquiry concerning this communication or earlier communications from the examiner should be directed to Benjamin W. Lee whose telephone number is 571-270-1346. The examiner can normally be reached on Mon - Fri (8:30 - 5:00).

If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Robert Pezzuto can be reached on 571-272-6996. The fax phone number for the organization where this application or proceeding is assigned is 571-273-8300.

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Benjamin W. Lee October 15, 2007

Ronald Laneau Primary Examiner Art Unit 3714

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10/15/07